

MOVEP 2012 Tutorial

Safety, Dependability and Performance Analysis of Extended AADL Models

Part 1: Overview



European Space Agency
European Space Research and Technology Centre



RWTH Aachen University
Software Modeling and Verification Group
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Fondazione Bruno Kessler
Centre for Scientific and Technological Research
Alessandro Cimatti

MOVEP 2012 School; December 7, 2012; Marseille, France

- ① Overview [Noll]
- ② System Modeling Using AADL [Noll]
- ③ Checking Functional Correctness [Cimatti]

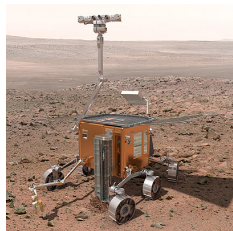
Coffee Break

- ④ Safety and Dependability Analysis [Cimatti]
- ⑤ Fault Detection, Isolation and Recovery (FDIR) Analysis [Cimatti]
- ⑥ Performability Evaluation [Noll]

Contents of Overview

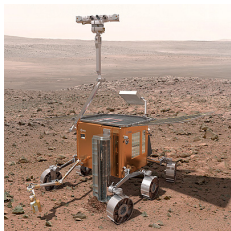
- 1 Introduction
- 2 COMPASS Project Overview
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ExoMars Rover: autonomy

- 4 to 21 min. for radio latency to earth
- infrequent communication opportunities (one or two short sessions per Martian day)



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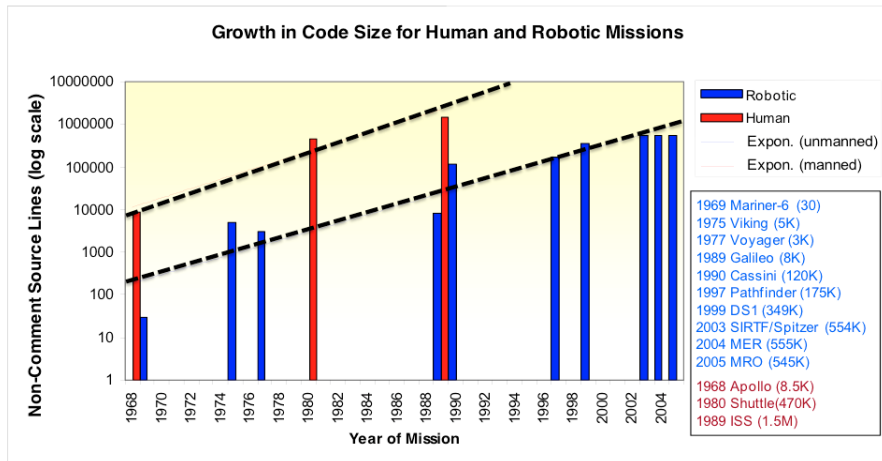
Autonomous Transfer Vehicle (ATV):

autonomy and **safety**

- fully-automated navigation and docking to ISS
 - human-rated requirements for safety (of ISS)
- ⇒ multi-failure tolerance (1 MLOC of control code)

© ESA/D. Barre - (2010)

Spacecraft = Flying Software

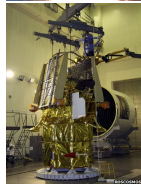


NASA Study on Flight Software Complexity (2009)

Extreme Dependability!

Requirements

- Must offer **service without interruption** for a very long time – typically years or decades
- **Faults** are costly and may severely damage reputations:
 - Ariane 5 crash in 1996 due to arithmetic overflow
 - Launch failure of recent Phobos-Grunt sample return mission
- “Five nines” (99.999 %) dependability **not** sufficient



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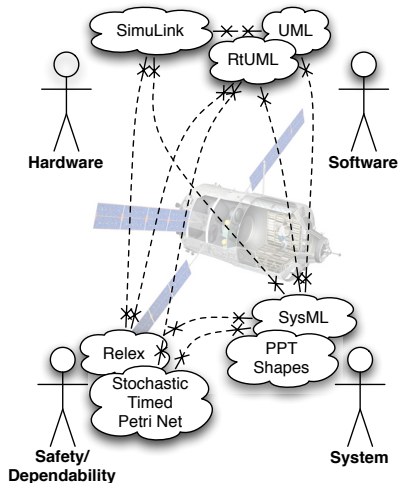
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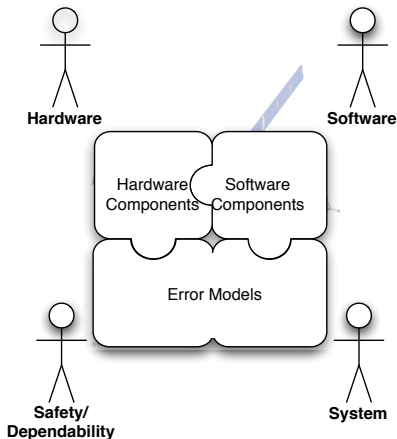
Challenges

- Rigorous **design support** and **analysis** techniques are called for
- Bugs must be found **as early as possible** in the design process
- Check **performance and reliability guarantees** whenever possible
- Effect of **Fault Detection, Isolation and Recovery** (FDIR) measures must be quantifiable



Limitations

- HW verified independently of SW with exaggerated mutual assumptions
- Safety & dependability analyses isolated from HW/SW models
- Multiple modeling formalisms for different system aspects (e.g. real-time, probabilistic, hybrid)
- No coherent approach to study effectiveness of FDIR



Solutions

Combination of

- HW, SW and their bindings +
- real-time, hybrid and probabilistic aspects +
- error models +
- non-nominal modes

in a **single integrated model**

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The COMPASS mission

Develop a **model-based** approach to **system-software co-engineering** while focusing on a **coherent set of modeling and analysis techniques** for evaluating system-level correctness, safety, dependability, and performance of **on-board computer-based aerospace systems**.



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Derived objectives

- 1 **Modeling formalism:** variant of AADL called SLIM (SAE **A**rchitecture **A**nalysis and **D**esign **L**anguage/**S**ystem-**L**evel **I**ntegrated **M**odeling Language)
- 2 **Verification methodology** based on state-of-the-art formal methods
- 3 **Toolset** supporting the analysis of AADL models
- 4 **Evaluation** on industrial-size case studies from aerospace domain

Consortium

- **RWTH Aachen University**
Software Modeling and Verification Group
- **Fondazione Bruno Kessler**
Embedded Systems Group
- **Thales Alenia Space**
World-wide #1 in satellite systems
- **Ellidiss Technologies**
AADL software tools

Funding & supervision

- **European Space Agency**



COMPASS Project Phases

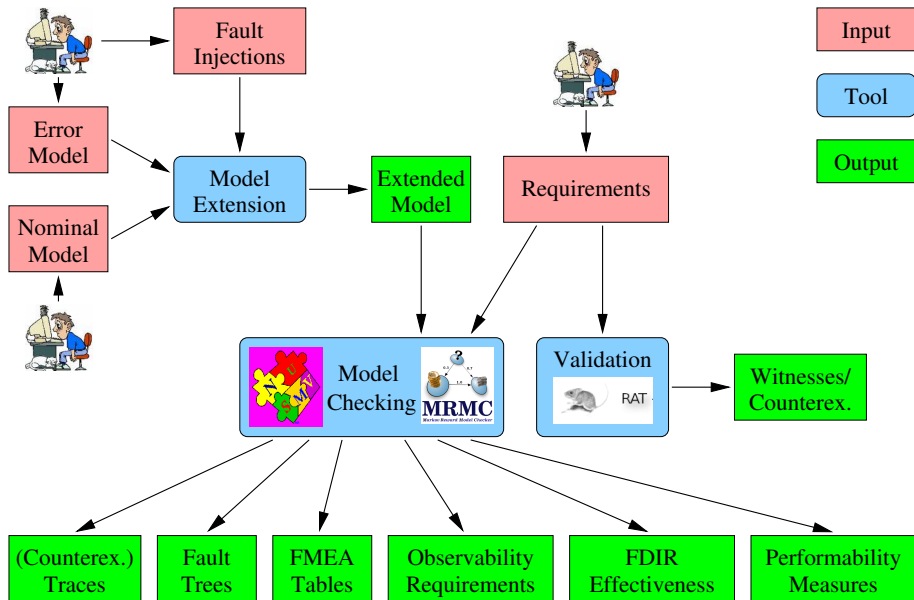
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|---|--|----------------------|
| ① | Project kick-off | February 2008 |
| ② | Language design | |
| ③ | Software tool specification + software design document | |
| ④ | Formal semantics | October 2008 |
| ⑤ | Prototype tool implementation | April 2009 |
| ⑥ | Prototype evaluation | |
| ⑦ | Final tool implementation | December 2009 |
| ⑧ | Final tool evaluation | March 2010 |
| ⑨ | Project extension | until March 2011 |
| ⑩ | New projects (NPI, CGM) | until September 2012 |
| ⑪ | Other application domains (D-MILS, HASDEL) | since November 2012 |

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Total budget: \approx 900 kEuro; \approx 10 programmers involved at peak times

COMPASS Methodology





- Symbolic LTL and CTL model checker
- BDD- and SAT-based model checking
- SMT-based timed model checking
- Counterexample generation



MRMC
Markov Reward Model Checker

- Model checker for MRMs
- Logics: PCTL and CSL (+rewards)
- Numerical + DES engine
- Bisimulation minimisation



RAT

- Requirements analyser
- Checks logical consistency

FSAP

- Safety analyser
- Fault-tree analysis

SigRef

- (MT)BDD bisimulation minimisation
- Models: Markov chains

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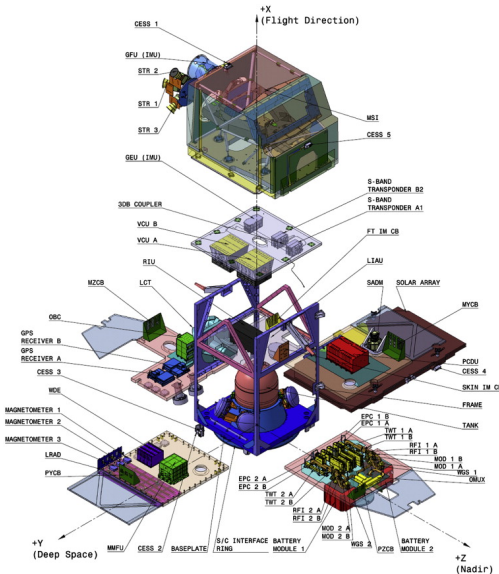
Case Study: Platform of

Launches between 2012-2020

Satellite

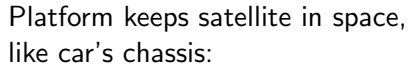
Platform keeps satellite in space, like car's chassis:

- control & data unit,
- propulsion,
- telemetry, tracking & cmd,
- power,
- attitude & orbit control sys,
- reconfiguration modules,
- etc.



Note: Shown picture is not from the case study

Satellite



- control & data unit,
- propulsion,
- telemetry, tracking & cmd,
- power,
- attitude & orbit control sys,
- reconfiguration modules,
- etc.

- redundancies + recovery,
- compensation algorithms,
- failure isolation schemes,
- omnipresent in satellite

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Verification & validation objectives

- Ensure that nominal and degraded conditions are correctly handled by **FDIR system**
- Ensure that **performance and risks** are within specified limits

AADL Model of Satellite Platform

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Model characteristics

✓ Functional	LOC (w/o comments):	3831	
✓ Probabilistic	Components:	86	Error models: 20
✓ Real-time	Ports:	937	Recoveries: 16
✓ Hybrid	Modes:	244	

State space of nominal behavior: **48,421,100** states

AADL Model of Satellite Platform

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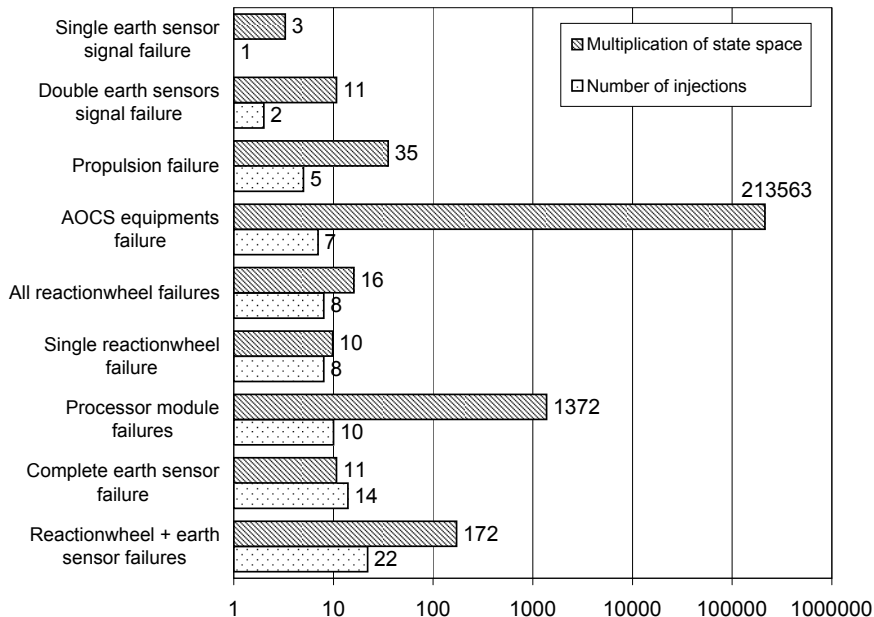
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Requirement metrics

- Functional properties: 42
(25 propositional, 2 absence, 1 universality, 14 response)
- Probabilistic properties: 2 (1 invariance, 1 existence)

State Space Growth by Fault Injection



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Achievements

- Component-based modeling framework based on AADL
- Novelties: dynamic reconfiguration, hybridity, error modeling, ...
- Automated correctness, safety, and performability analysis
- Industrial evaluation by third-party company showed maturity

Trustworthy aerospace design = AADL modeling + analysis

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Further information

- General approach (Yushstein et. al, [IEEE SMC-IT 2011](#))
(Bozzano et. al, [ACES-MB 2009](#))
(Bozzano et. al, [SAFECOMP 2009](#))
- AADL model checker (Bozzano et. al, [CAV 2010](#))
- Thales case studies (Bozzano et. al, [ERTS² 2010](#))
- ESA satellite case study (Esteve et. al, [ICSE 2012](#))
- Tool download at <http://compass.informatik.rwth-aachen.de/>